



Nickel Alloy - AMS 5663 - Alloy 718

SPECIFICATIONS

Aerospace	AMS 5663
Commercial	Alloy 718

Alloy 718 is a Nickel-Chromium based Superalloy

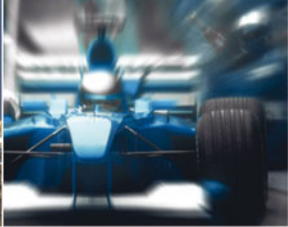
CHEMICAL COMPOSITION

Element	% Present
Nickel (Ni)	50 - 55
Nickel + Cobalt (Ni+Co)	50 - 55
Chromium (Cr)	17 - 21
Niobium (Columbium) (Nb)	4.75 - 5.5
Columbium + Tantalum (Cb+Ta)	4.75 - 5.5
Molybdenum (Mo)	2.8 - 3.3
Titanium (Ti)	0.65 - 1.15
Cobalt (Co)	1 max
Aluminium (Al)	0.2 - 0.8
Manganese (Mn)	0.35 max
Silicon (Si)	0.35 max
Copper (Cu)	0.3 max
Phosphorous (P)	0.15 max
Sulphur (S)	0.15 max
Carbon (C)	0.08 max
Boron (B)	0.06 max
Iron (Fe)	Balance

This details the main elements only

SUPPLIED FORMS

Please contact us with your requirements



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MECHANICAL PROPERTIES

<i>Typical</i>	
Property	Value
Proof Stress	70 MPa
Tensile Strength	135 MPa
Elongation A50 mm	45 %
Hardness Rockwell B	100 HRB

The table shows typical properties for this alloy

PHYSICAL PROPERTIES

Density	0.296 lb/in ³ annealed	0.274 lb/in ³ aged						
Melting Point	2410 - 2540 °F							
Temperature, °F	-320	70	200	400	600	1000	1200	1400
Coefficient of Thermal Expansion, in/in °F x 10 ⁻⁴	5.9	-	7.3	7.5	7.7	8.1	8.4	8.9
Thermal Conductivity Btu ft/ft ²	-	6.4	7.2	8.2	9.3	11.3	12.3	13.3
Modulus of Elasticity, Dynamic psi x 10 ⁶	-	29	28	27	26	25	24	22



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CONTACT

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REVISION HISTORY

Datasheet Updated	15 January 2019
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This Data is indicative only and as such is not to be relied upon in place of the full specification. In particular, mechanical property requirements vary widely with temper, product and product dimensions. All information is based on our present knowledge and is given in good faith. No liability will be accepted by the Company in respect of any action taken by any third party in reliance thereon.

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